

T (°C)	P (kPa)	v (m³/kg)	تاز
-8	320	0.0007571	مايع متكاثف ✓
30	770.64	0.015	مخلوط
-12.73	180	0.11041	بخار اشباع
80	600	0.044710	بخار فوق اشباع

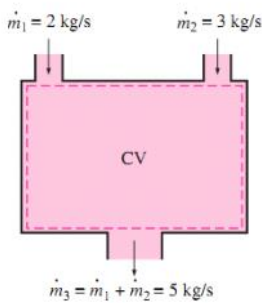
$$\frac{R-13 \text{ E a}}{-2. \text{ c}}$$

$$\frac{R-22}{-1. \text{ c}}$$

$$\begin{cases} T = -1 \\ P = 320 \text{ kPa} \end{cases} \rightarrow T < T_{sat} \rightarrow T_{sat} = 2.1 \text{ E c}$$

لبسبم باز  
در داخل سيستم جرم يا انرژي ذخيره نمائيم  
در داخل سيستم جرم يا انرژي لاگولنه ذخيره نمائيم

اولين صورت: صورت جرمي



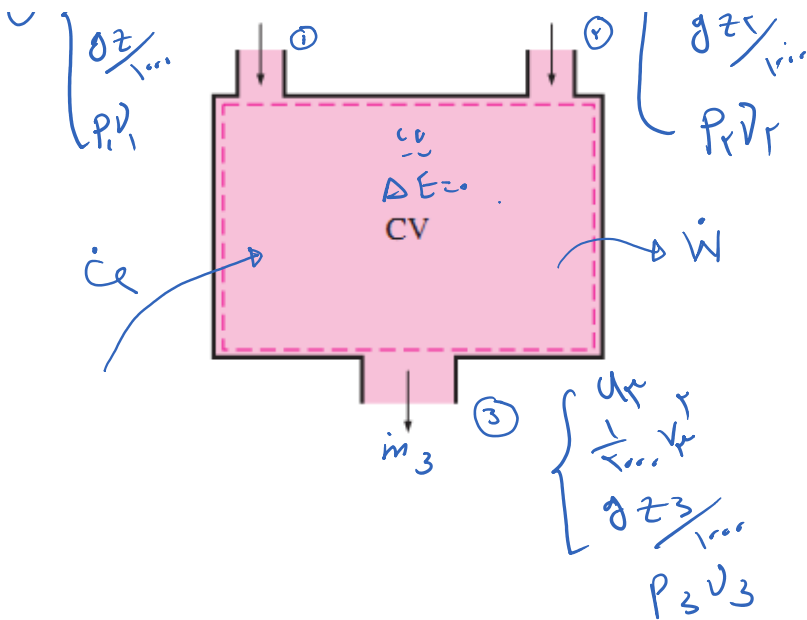
$$\dot{m}_{in} = \dot{m}_{out} \Rightarrow \dot{m}_1 + \dot{m}_2 = \dot{m}_3$$

$$2 + 3 = 5 = \dot{m}_3 \text{ (kg/s)}$$

$$h = u + Pv$$

$$\dot{m} = \rho Q = \rho A V$$



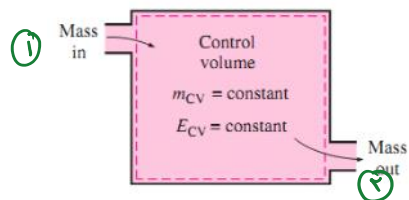


$$\Delta E = 0 \rightarrow E_{in} = E_{out} \rightarrow \dot{Q} + \dot{m}_1 \left[ h_1 + \frac{1}{\rho_1} v_1^2 + \frac{g z_1}{1000} \right] + \dot{m}_2 \left[ h_2 + \frac{v_2^2}{\rho_2} + \frac{g z_2}{1000} \right] = \dot{W} + \dot{m}_3 \left[ h_3 + \frac{v_3^2}{\rho_3} + \frac{g z_3}{1000} \right]$$

$$\dot{Q} - \dot{W} = \dot{m}_3 \left[ h_3 + \frac{v_3^2}{\rho_3} + \frac{g z_3}{1000} \right] - \dot{m}_1 \left[ h_1 + \frac{v_1^2}{\rho_1} + \frac{g z_1}{1000} \right] - \dot{m}_2 \left[ h_2 + \frac{v_2^2}{\rho_2} + \frac{g z_2}{1000} \right]$$

زمانی که در یک سیستم دو ورودی و یک خروجی معلوم داریم

$$\dot{m}_{in} = \dot{m}_{out} \rightarrow \dot{m}_1 = \dot{m}_2$$



$$\dot{Q} - \dot{W} = \dot{m}_2 \left[ h_2 + \frac{v_2^2}{\rho_2} + \frac{g z_2}{1000} \right] - \dot{m}_1 \left[ h_1 + \frac{v_1^2}{\rho_1} + \frac{g z_1}{1000} \right]$$

$$\dot{Q} - \dot{W} = \dot{m} \left[ h_2 - h_1 + \frac{v_2^2 - v_1^2}{\rho} + \frac{g(z_2 - z_1)}{1000} \right]$$

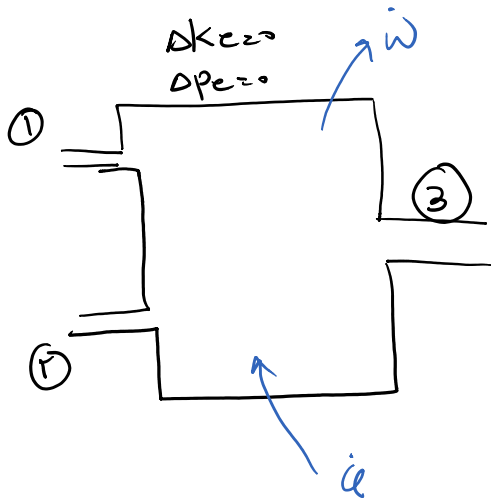
$$\Delta ke = 0$$

$$\Delta pe = 0$$



$$\dot{Q} - \dot{W} = \dot{m}(h_r - h_1)$$

$$q - w = h_r - h_1$$



$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2$$

$$\dot{Q} - \dot{W} = \dot{m}_3 h_3 - \dot{m}_1 h_1 - \dot{m}_2 h_2$$

**EXAMPLE 5-4 Deceleration of Air in a Diffuser**

Air at 10°C and 80 kPa enters the diffuser of a jet engine steadily with a velocity of 200 m/s. The inlet area of the diffuser is 0.4 m<sup>2</sup>. The air leaves the diffuser with a velocity that is very small compared with the inlet velocity. Determine (a) the mass flow rate of the air and (b) the temperature of the air leaving the diffuser.

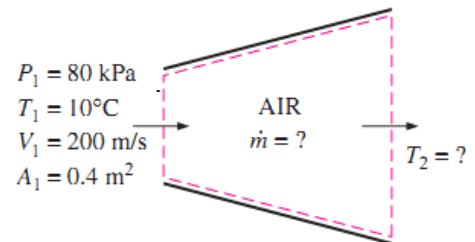
Air

$$\textcircled{1} \begin{cases} T = 10^\circ\text{C} \\ p = 80 \text{ kPa} \\ V = 200 \text{ m/s} \\ A = 0.4 \text{ m}^2 \end{cases}$$

$$\textcircled{2} \begin{cases} V = 0 \\ \dot{m} = ? \\ T = ? \end{cases}$$

$$h \longrightarrow T$$

$$T_1 = 283 \longrightarrow h_1 = 281.04 \text{ kJ/kg}$$



$$\dot{m}_{in} = \dot{m}_{out}$$

$$\dot{m}_{in} = \rho_{in} \cdot \dot{V}_{in} = \rho_{in} \cdot V_{in} \cdot A_{in} = \frac{1}{v_{in}} \cdot A_{in} \cdot V_{in} = \frac{p_1 \cdot x \cdot \epsilon}{1.013}$$

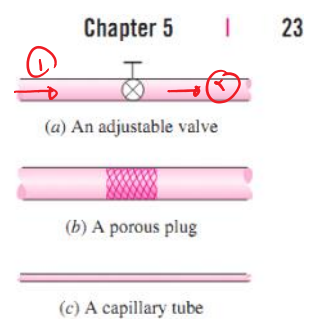
$$p_1 v_1 = R T_1 \Rightarrow 1.013 \cdot v_1 = 0.287 \cdot (283) \Rightarrow v_1 = 1.013$$

$$P_1 v_1 = R T_1 \Rightarrow \rho_1 \times v_1 = \frac{1}{\rho_1} \times \rho_1 \times (R T_1) \rightarrow \rho_1 = 1018$$

$$q - w = (h_r - h_i) + \frac{1}{\rho_{\dots}} (v_r^f - v_i^f) + \frac{g}{T_{\dots}} (z_r - z_i)$$

$$h_r - h_i - \frac{v_i^f}{\rho_{\dots}} = 0 \rightarrow h_r - h_i = \frac{v_i^f}{\rho_{\dots}}$$

$$h_r = \frac{c \times 10^6}{\rho_{\dots}} + 213 = \frac{c}{\rho} + 213 = 413 \rightarrow T_2 = 210 \text{ K}$$



فرانیدر سیدرمان فرگسین آن کابله ثابت است.

$$q - w = (h_r - h_i) + \frac{1}{\rho_{\dots}} (v_r^f - v_i^f) + \frac{g}{T_{\dots}} (z_r - z_i)$$

$$\Rightarrow h_r = h_i$$

**EXAMPLE 5-8 Expansion of Refrigerant-134a in a Refrigerator**  
 Refrigerant-134a enters the capillary tube of a refrigerator as saturated liquid at 0.8 MPa and is throttled to a pressure of 0.12 MPa. Determine the quality of the refrigerant at the final state and the temperature drop during this process.

① {  
 طبع (ب) ع : فزیدل  
 $P = 0.12 \text{ MPa} = 120 \text{ kPa}$

② {  
 $P = 0.8 \text{ MPa}$   
 $x = ?$   
 $\Delta T = ?$

22.49 214.48 236.97

$h_g = 234.14$

① {  
 $\rho_1 = \rho_f = 1018$   
 $T_1 = 313.1$   
 $s_1 = 1.3242$

② {  
 $P = 0.12 \text{ MPa} = 120 \text{ kPa} \rightarrow h_g = 211.29$   
 $h_r = h_i = 98.14$

$$\textcircled{1} \begin{cases} T_1 = 313 \text{ K} \\ S_1 = 158.8 \text{ J/K} \\ h_1 = 98.1 \text{ kJ} \end{cases}$$

$$\Delta T_2 T_2 - T_1 = -22.22 - 313$$

$$= -335.22$$

$$\textcircled{2} \begin{cases} h_r = h_i = 98.1 \text{ kJ} \\ h_f < h_r < h_g \rightarrow \\ T_2 = 22.22 \end{cases}$$

مختلط

$$h_r = h_f + x(h_g - h_f)$$

$$98.1 \text{ kJ} = 22.22 + x(212.1) \rightarrow x = \frac{98.1 \text{ kJ} - 22.22}{212.1}$$

$$x = 0.35$$